



stiffness.

If the virtual constraints used in the walking experiment were applied to the stance phase of running, the same set of springs would yield stance times of around 100 ms. Further, since there is no control authority on the torso in the flight phase, due to the conservation of angular momentum, any tracking error on the torso position has to be corrected during the stance phase. Feedback to correct the possibly large errors for the torso within the short stance phase would place large torque requirements on the actuators, and would potentially be infeasible. Hence, longer stance times are necessary.

One solution to obtain longer stance times would be to reduce the spring stiffness by physically replacing the springs present in MABEL with softer springs. However, as investigated in Rummel and Seyfarth [20], having compliance in the joint level with segmented legs results in a nonlinear relationship between leg compression and leg force. Softer springs would lead to the robot collapsing at moderate leg compressions owing to the fact that the less-stiff spring is unable to provide sufficient leg force to hold up the robot. This would significantly reduce the range of impact angles for the knee for which the springs could support the weight of the robot. Thus, there is a need to vary the effective compliance of the leg in different parts of the stance phase without resorting to softer springs.

For inspiration, we take a look at a few biomechanical studies. Ferris et al., [4], [5] carried out experiments on human runners and found that runners adjust their leg stiffness to accommodate for variations in surface stiffness, allowing them to maintain similar running mechanics (e.g., peak ground reaction force and ground contact time) on different surfaces. Moreover, they suggest that incorporating an adjustable leg stiffness in the design of running robots is important if they are to match the agility and speed of animals on varied terrain. Moreover, in a set of impressive experiments carried out by Daley et al., [3], [2], where guinea fowl are subjected to large unexpected variations in ground terrain, it is suggested that the animals can accommodate this variation in ground height by varying their leg stiffness.

Furthermore, active force control has been suggested as a way to increase robustness to perturbations in ground height and ground stiffness in [11].

In summary, there is a need for a control strategy which can dynamically vary the effective compliance of the leg. In this paper, we present a controller based on virtual constraints and the framework of hybrid zero dynamics to create a zero dynamics that is both compliant and actuated. The actuator within the compliant hybrid zero dynamics is utilized to implement an active force control strategy so as to have the capability of dynamically varying the effective leg stiffness. With this in mind, Section II presents a mathematical model for MABEL, Section III presents the technical details for the control design, Section IV presents an experimental validation of the running controller, and Section V presents a few concluding remarks.

## II. MABEL'S MODEL

This section develops the hybrid model appropriate for a running gait comprised continuous phases representing stance and flight phases of running, and discrete transition between the two. Standard model hypotheses for a running gait and rigid impact as in [26, pp. 50–51] are assumed. In particular, the stance phase is a single support phase with one foot assumed pinned to the ground while the flight phase has both feet above the ground. The stance to flight transition is usually a trivial lift map [26], however for MABEL, due to the unilateral spring, this transition models an internal impact of the spring with a hardstop (see Figure 1.) On the other hand, the flight to stance transition models an instantaneous rigid impact, representing the impact of the swing toe with the ground. Both impacts models are based on [6].

### A. MABEL's Unconstrained Dynamics

The configuration space  $Q_e$  of the unconstrained dynamics of MABEL is an open simply-connected subset of  $S^7 \times \mathbb{R}^2$ : five DOF are associated with the links in the robot's body, two DOF are associated with the springs in series with the two leg-shape motors, and two DOF are associated with the horizontal and vertical position of the robot in the sagittal plane. A set of coordinates suitable for parametrization of the robot's linkage and transmission is,  $q_e := (q_{LA_{st}}; q_{mLS_{st}}; q_{BSP_{st}}; q_{LA_{sw}}; q_{mLS_{sw}}; q_{BSP_{sw}}; q_{Tor}; p_{hip}^h; p_{hip}^v)$ , where, as in Figure 1,  $q_{Tor}$  is the torso angle, and  $q_{LA_{st}}$ ,  $q_{mLS_{st}}$ , and  $q_{BSP_{st}}$  are the leg angle, leg-shape motor position and  $B_{spring}$  position respectively for the stance leg. The swing leg variables,  $q_{LA_{sw}}$ ,  $q_{mLS_{sw}}$  and  $q_{BSP_{sw}}$  are defined similarly. For each leg,  $q_{LS}$  is uniquely determined by a linear combination of  $q_{mLS}$  and  $q_{BSP}$ , reflecting the fact that the cable differentials place the spring in series with the motor, with the pulleys introducing a gear ratio. The coordinates  $p_{hip}^h, p_{hip}^v$  are the horizontal and vertical positions of the hip in the sagittal plane.

The equations of motion are obtained using the method of Lagrange. In computing the Lagrangian, the total kinetic energy is taken to be the sum of the kinetic energies of the transmission, the rigid linkage, and the boom. The potential energy is computed in a similar manner with the difference being that the transmission contributes to the potential energy of the system only through its non-elastic energy (the mass). This distinction is made since it is more convenient to model the unilateral spring as an external input to the system. The resulting model of the robot's unconstrained dynamics is determined as

$$D_e(q_e) \ddot{q}_e + C_e(q_e, \dot{q}_e) \dot{q}_e + G_e(q_e) = \Gamma_e, \quad (1)$$

where,  $D_e$  is the inertia matrix,  $C_e$  contains Coriolis and centrifugal terms,  $G_e$  is the gravity vector, and  $\Gamma_e$  is the vector of generalized forces acting on the robot, expressed as,

$$\Gamma_e = B_e u + E_{ext}(q_e) F_{ext} + B_{fric} \tau_{fric}(q_e, \dot{q}_e) + B_{sp} \tau_{sp}(q_e, \dot{q}_e), \quad (2)$$

where the matrices  $B_e$ ,  $E_{\text{ext}}$ ,  $B_{\text{fric}}$ , and  $B_{\text{sp}}$  are derived from the principle of virtual work and define how the actuator torques  $u$ , the external forces  $F_{\text{ext}}$  at the leg, the joint friction forces  $\tau_{\text{fric}}$ , and the spring torques  $\tau_{\text{sp}}$  enter the model respectively. The dimension of  $u$  is four, corresponding to the two actuators on each leg for actuating leg shape and leg angle.

### B. MABEL's Constrained Dynamics

The model (1) can be particularized to describe the stance and flight dynamics by incorporating proper holonomic constraints.

1) *Dynamics of Stance*: For modeling the stance phase, the stance toe is assumed to act as a passive pivot joint (no slip, no rebound and no actuation). Hence, the Cartesian position of the hip,  $(p_{\text{hip}}^h, p_{\text{hip}}^v)$ , is defined by the coordinates of the stance leg and torso. The springs in the transmission are appropriately chosen to support the entire weight of the robot, and hence are stiff. Consequently, it is assumed that the spring on the swing leg does not deflect, that is,  $q_{\text{Bsp}_{\text{sw}}} \equiv 0$ . The stance configuration space,  $Q_s$ , is therefore a co-dimension three submanifold of  $Q_e$ . With these assumptions, the generalized configuration variables in stance are taken as  $q_s := (q_{\text{LA}_{\text{st}}}; q_{\text{mLS}_{\text{st}}}; q_{\text{Bsp}_{\text{st}}}; q_{\text{LA}_{\text{sw}}}; q_{\text{mLS}_{\text{sw}}}; q_{\text{Tor}})$ . Defining the state vector  $x_s := (q_s; \dot{q}_s) \in TQ_s$ , the stance dynamics can be expressed in standard form as,

$$\dot{x}_s = f_s(x_s) + g_s(x_s)u. \quad (3)$$

2) *Dynamics of Flight*: In the flight phase, both feet are off the ground, and the robot follows a ballistic motion under the influence of gravity. Thus the flight dynamics can be modeled by the unconstrained dynamics developed earlier. However an additional assumption can be made to eliminate the stiffness in integrating the differential equations representing the flight model. As mentioned, the springs must be stiff to support the entire weight of the robot. Further, since neither leg is in contact with the ground during the flight phase, it can be assumed that the springs on each leg do not deflect during the flight phase. Therefore,  $q_{\text{Bsp}_{\text{st}}} \equiv 0, q_{\text{Bsp}_{\text{sw}}} \equiv 0$ . Thus, the configuration space of the flight dynamics is a co-dimension two submanifold of  $Q_e$ , i.e.,  $Q_f := \{q_e \in Q_e \mid q_{\text{Bsp}_{\text{st}}} \equiv 0, q_{\text{Bsp}_{\text{sw}}} \equiv 0\}$ . It follows that the generalized configuration variables in the flight phase can be taken as  $q_f := (q_{\text{LA}_{\text{st}}}; q_{\text{mLS}_{\text{st}}}; q_{\text{LA}_{\text{sw}}}; q_{\text{mLS}_{\text{sw}}}; q_{\text{Tor}}; p_{\text{hip}}^h; p_{\text{hip}}^v)$ . Defining the state vector  $x_f := (q_f; \dot{q}_f) \in TQ_f$ , the flight dynamics can be expressed in standard form as,

$$\dot{x}_f = f_f(x_f) + g_f(x_f)u. \quad (4)$$

### C. MABEL's Transitions

1) *Stance to Flight Transition Map*: Physically, the robot takes off when the normal component of the ground reaction force acting on the stance toe,  $F_{\text{toe}_{\text{st}}}^N$ , becomes zero. The ground reaction force at the stance toe can be computed as a function of the acceleration of the COM and thus depends on the inputs  $u \in \mathcal{U}$  of the system described by

(3). Mathematically, the transition occurs when the solution of (3) intersects the co-dimension one switching manifold

$$\mathcal{S}_{\text{s} \rightarrow \text{f}} := \{x_s \in TQ_s \times \mathcal{U} \mid F_{\text{toe}_{\text{st}}}^N = 0\}. \quad (5)$$

On transition from the stance to flight phase, the stance leg comes off the ground and takeoff occurs. During the stance phase, the spring on the stance leg is compressed. When the stance leg comes off the ground, the spring rapidly decompresses and impacts the hardstop. The stance to flight transition map,  $\Delta_{\text{s} \rightarrow \text{f}} : \mathcal{S}_{\text{s} \rightarrow \text{f}} \rightarrow TQ_f$  accounts for this. Further details are omitted for the sake of brevity and interested readers are referred to [21, Ch. III].

2) *Flight to Stance Transition Map*: The robot physically transitions from flight phase to stance phase when the swing toe contacts the ground surface. The impact is modeled here as an inelastic contact between two rigid bodies. It is assumed that there is no rebound or slip at impact. Mathematically, the transition occurs when the solution of (4) intersects the co-dimension one switching manifold

$$\mathcal{S}_{\text{f} \rightarrow \text{s}} := \{x_f \in TQ_f \mid p_{\text{toe}_{\text{sw}}}^v = 0\}. \quad (6)$$

In addition to modeling the impact of the leg with the ground, and the associated discontinuity in the generalized velocities of the robot [6], the transition map accounts for the assumption that the spring on the new swing leg remains at its rest length, and for the relabeling of the robot's coordinates so that only one stance model is necessary. In particular, the transition map consists of three subphases executed in the following order: (a) standard rigid impact model [6]; (b) adjustment of spring velocity in the new swing leg; and (c) coordinate relabeling.

The flight to stance transition map,  $\Delta_{\text{f} \rightarrow \text{s}} : \mathcal{S}_{\text{f} \rightarrow \text{s}} \rightarrow TQ_s$ , is similar to the one developed in [16, Ch. V], [21, Ch. III] and further details are omitted for the sake of brevity.

### D. Hybrid model of Running

The hybrid model of running is based on the dynamics developed in Section II-B and the transition maps presented in Section II-C, and is given by

$$\Sigma_s : \begin{cases} \dot{x}_s = f_s(x_s) + g_s(x_s)u, & (x_s^-, u^-) \notin \mathcal{S}_{\text{s} \rightarrow \text{f}} \\ x_f^+ = \Delta_{\text{s} \rightarrow \text{f}}(x_s^-, u^-), & (x_s^-, u^-) \in \mathcal{S}_{\text{s} \rightarrow \text{f}} \end{cases} \quad (7)$$

$$\Sigma_f : \begin{cases} \dot{x}_f = f_f(x_f) + g_f(x_f)u, & x_f^- \notin \mathcal{S}_{\text{f} \rightarrow \text{s}} \\ x_s^+ = \Delta_{\text{f} \rightarrow \text{s}}(x_f^-), & x_f^- \in \mathcal{S}_{\text{f} \rightarrow \text{s}}. \end{cases}$$

## III. CONTROL DESIGN FOR RUNNING

This section presents a controller design for inducing stable running motions on MABEL. The controller will create an actuated compliant HZD enabling active force control within the HZD.

Virtual constraints for the stance phase of running are chosen in a manner similar to that of walking [24] such that the open-loop compliance of the system is preserved as a dominant characteristic of the closed-loop system. However, by implementing one less virtual constraint in the stance



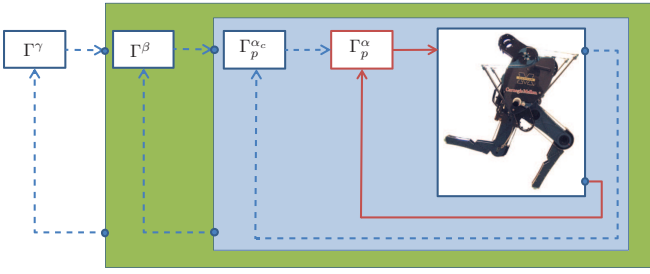


Fig. 2. Feedback diagram illustrating the running controller structure. Continuous lines represent signals in continuous time; dashed lines represent signals in discrete time. The controllers  $\Gamma_p^\alpha$  and  $\Gamma_p^{\alpha_c}$  create a compliant actuated hybrid zero dynamics. The controller  $\Gamma^\beta$  ensures that the periodic orbit on the resulting zero dynamics manifold is locally exponentially stable. The controller  $\Gamma^\gamma$  improves the domain of attraction of the periodic orbit.

phase than the maximum possible, an actuator is left free and will result in the zero dynamics being actuated. Through this actuator, active force control will be introduced as a means of varying the effective compliance of the system. The rest of the section is organized as follows. Section III-A presents a high-level overview of the control design. Section III-B presents the virtual constraints. Section III-C presents one fixed point representing a periodic running motion. Section III-D presents the closed-loop design.

#### A. Overview of the Control Method

The control objective is to design a periodic running gait that is exponentially stable and has a sufficiently large domain of attraction so as to accommodate inevitable differences between the model and the robot. Virtual constraints are used to synchronize the robot's links throughout the stance and flight phases. By a judicious choice of variables on which the constraints are to be imposed, the resulting restricted stance dynamics is made compliant and actuated. The input in the zero dynamics for the stance phase is used to change the effective compliance of the robot. Discrete-event-based control is then employed to (a) create hybrid invariance, (b) exponentially stabilize the periodic gait, and (c) increase the domain of attraction of the periodic gait.

To achieve the control objectives, the feedback controller introduces control on three levels. Figure 2 depicts the overall structure of the running controller. On the first level, continuous-time feedback controllers  $\Gamma_p^\alpha$  with  $p \in \mathcal{P} := \{s, f\}$  are employed in the stance and flight phases to create invariant and attractive surfaces embedded in the state space for each of the respective phases. The discrete-time feedback controllers  $\Gamma_p^{\alpha_c}$  are employed in the transitions between the phases in order to render these surfaces hybrid invariant.

On the second level, an event-based controller  $\Gamma^\beta$  performs step-to-step parameter updates to render the periodic orbit, representing running and embedded in these surfaces, exponentially stable. Finally, on the third level, another event-based controller  $\Gamma^\gamma$  performs step-to-step parameter updates to increase the domain of attraction of the periodic orbit.

The remaining sections of this chapter will describe the procedure detailed above in greater detail.

#### B. Virtual Constraint Design and Active Force Control

Virtual constraints [26] are holonomic constraints that are parametrized by a monotonic function of the state and imposed through feedback control, with the purpose being to restrict the dynamics to evolve on lower-dimensional surfaces embedded in the state spaces of the stance and flight dynamics. This lower-dimensional hybrid system governs the existence and the stability of periodic solutions corresponding to running motions. The virtual constraints for running can be described by a choice of outputs,

$$y_p = H_0^p q_p - h_d^p(\theta_p(x_p), \alpha_p, \alpha_c^p, \beta, \gamma), \quad (8)$$

where  $p \in \mathcal{P}$ ,  $h_d^p$  is the desired evolution of the virtual constraints which is parametrized by Bézier polynomials with coefficients  $\alpha_p$ . The other Bézier polynomial coefficients,  $\alpha_c^p$ ,  $\beta$ , and  $\gamma$  are zero for the nominal gait and are updated in an event-based manner. In particular,  $\alpha_c^p$  parametrize correction polynomials that are used to create hybrid invariance, while  $\beta$  and  $\gamma$  are used by outer-loop event-based controllers to make step-to-step updates to the virtual constraints.

For the stance phase,  $H_0^s$  is based on the walking controller introduced in [24], but with the stance motor leg shape variable omitted. A virtual constraint on the torso position provides a desired profile for the torso, and two virtual constraints on the swing leg angle and the swing motor leg shape describe the evolution of the swing leg. With the choice of these three virtual constraints, the stance zero dynamics results in being both compliant and actuated. Further details are given in [21, Ch. VI] and are omitted here for brevity. The stance virtual constraints are parametrized as a function of  $\theta_s$ , shown in Figure 1.

The stance leg shape motor is the actuator that moves into the zero dynamics. Due to the transmission in MABEL, this actuator is in series with the spring. By imposing a torque of the form  $u_{mLS_{st}} = -k_{vc}(q_{mLS_{st}} - q_{mLS_{vc}})$  on this actuator, a virtual compliant element with stiffness  $k_{vc}$  and rest position  $q_{mLS_{vc}}$  is created and placed in series with the physical compliance. This active force control strategy enables changing the effective compliance of the stance leg dynamically. However, to keep the controller simple, the virtual compliance parameters are modified only once during the stance phase. The stance phase is artificially divided into stance-compression (*sc*) and stance-decompression (*sd*) subphases, and the parameters for the virtual compliance are updated only at this transition.

For the flight phase,  $H_0^f$  is chosen as follows. On the stance leg, the leg angle and motor leg shape variables are chosen, and on the swing leg, the absolute leg angle and motor leg shape variables are chosen. The absolute leg angle on the swing leg enables directly specifying the touchdown angle through a virtual constraint. The flight virtual constraints are parametrized as a function of  $\theta_f$ , which is chosen as the horizontal position of the hip, as in RABBIT [13].

The choice of the desired evolution of the virtual constraints,  $h_d^p$  for the stance and flight phases, and the choice of the virtual compliance for the stance-compression and

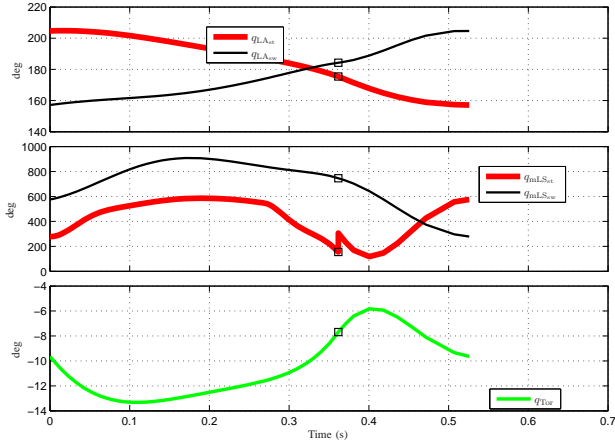


Fig. 3. Evolution of the virtual constraints and configuration variables for a nominal fixed point (periodic running gait) at a speed of 1.34 m/s and step length 0.7055 m. The squares illustrate the location of transition between stance to flight phase.

stance-decompression subphases is left as a free parameter to be found by optimization.

### C. Fixed Point for Running

A periodic running gait is designed by selecting the free parameters in the virtual constraints and the virtual compliance by posing and solving a constrained optimization problem (see [26, Ch. 6]). A nominal fixed point representing running at 1.34 m/s was obtained with a step time of 525 ms, with 69% spent in stance and 31% in flight. Figures 3-7 illustrate various variables for the nominal fixed point. In all of these figures, the squares on the plots indicate the location of the transition from stance to flight phase.

Figure 3 illustrates the nominal evolution of the virtual constraints for the stance and flight phases along with other configuration variables for one step of running. The instantaneous change in the stance motor leg shape position on transition to flight is to reset the stance spring to its rest position in the flight phase.

Figure 4 illustrates the evolution of the leg shape and the stance  $B_{\text{spring}}$  variables. The circle in the spring plot indicates the location of stance-compression to stance-decompression transition. During the flight phase, the stance leg shape initially unfolds due to the large velocity of push-off during the final part of the stance phase as the spring rapidly decompresses. During the stance-compression phase, the spring compresses, reaches its peak value of almost  $36^\circ$ , and starts to decompress. On transition to the stance-decompression phase, a change in the virtual compliance parameters causes the motor to inject energy into the system, causing the spring to rapidly compress to a peak of  $47^\circ$ . At lift-off, when the vertical component of the ground force goes to zero (see Figure 5), the spring has decompressed to approximately  $25^\circ$ .

Figure 6 illustrates the evolution of the swing leg height and the vertical position of the center of mass of the robot. The swing foot is over 15 cm above the ground at its peak

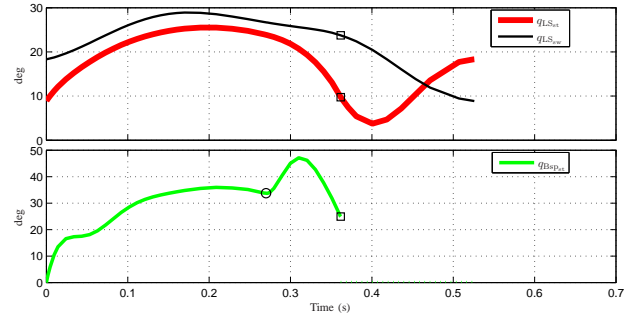


Fig. 4. Evolution of the leg shape and stance  $B_{\text{spring}}$  variables corresponding to the nominal fixed point. The squares illustrate the location of transition between stance to flight phase. The circle on the  $B_{\text{spring}}$  plot illustrates the location of the sc to sd event transition.

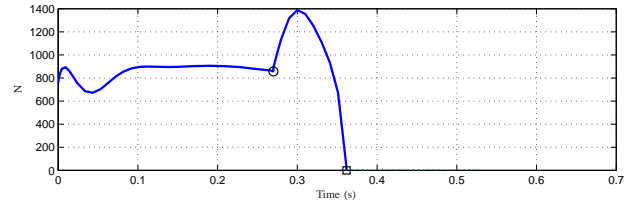


Fig. 5. Vertical component of the ground reaction force for the nominal running fixed point. At the sc to sd event transition (indicated by the circle), the change in the offset for the virtual compliance causes the spring to compress further which increases the ground reaction force considerably. Takeoff occurs when the ground reaction force goes to zero (indicated by the square.)

to offer good ground clearance for hard impacts. During the stance phase, the COM undergoes an asymmetric motion with the lowest point of potential energy being around 52% into the stance phase. During the flight phase, the COM has a ballistic trajectory. Both these motions are dominant characteristics of running.

Figure 7 illustrates the actuator torques used to realize the gait. The stance and swing leg angle torques and the swing leg shape torque are small compared to the peak torque capacities of the actuators: 30Nm. The stance leg shape torque is large, initially to support the weight of the robot as

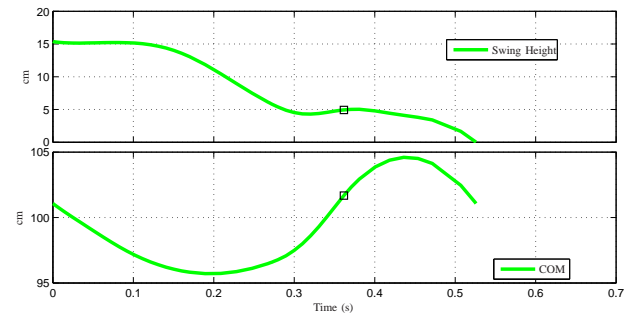


Fig. 6. Evolution of swing leg height and vertical center of mass (COM) of the robot for the nominal fixed point. The COM trajectory clearly illustrates the lowest point of potential energy during the stance phase and the ballistic trajectory in the flight phase, both of which are dominating characteristics of running. The squares illustrate the location of transition between stance to flight phase.

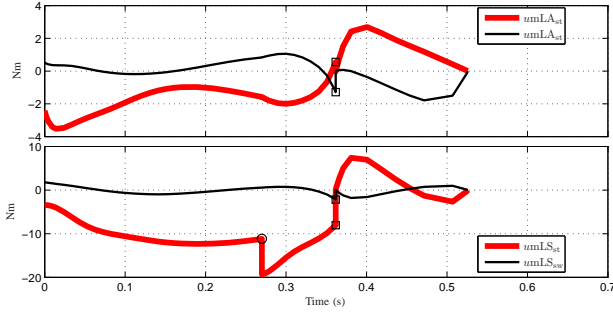


Fig. 7. Actuator torques corresponding to the nominal fixed point. The squares illustrate the location of transition between stance to flight phase. The circle on the  $umL_{S_{st}}$  plot illustrates the location of the  $sc$  to  $sd$  event transition. Note that the torques are discontinuous at stance to flight transitions. Also note the additional discontinuity for  $umL_{S_{st}}$  at the  $sc$  to  $sd$  event transition due to the instantaneous change in the offset for the virtual compliance at this transition.

the stance knee bends and subsequently to sufficient energy injection in the stance-decompression phase to achieve lift-off. The stance motor leg shape torque is discontinuous at the stance-compression to stance-decompression transition due to an instantaneous change in the parameters for the virtual compliance. All torques are discontinuous on the stance to flight transition due to the impact of the spring with the hard-stop.

#### D. Closed-loop Design and Stability Analysis

The periodic running motion in the previous section was found by studying the restricted hybrid dynamics of the system. We now need to design a controller that creates the lower-dimensional surfaces and makes them invariant and attractive. In the following, we introduce control action on three levels with an inner-loop and two outer-loops. On the first level, a continuous-time controller is presented that in addition to rendering the zero dynamics invariant also makes it attractive. The hybrid invariance is still achieved through the correction polynomials on a event to event level [12]. On the second level, an outer-loop event-based discrete linear controller is introduced to exponentially stabilize the periodic orbit representing the running gait. Finally on the third level, an additional outer-loop event-based nonlinear controller is introduced to enlarge the domain of attraction of the periodic orbit.

The classic input-output linearizing controller [26, Ch. 5] is used as  $\Gamma_p^\alpha$  to render the zero dynamics both invariant and attractive. The correction polynomials create hybrid invariance and are updated step-to-step by  $\Gamma_p^{\alpha_c}$ . The stability of the fixed point under the above controller can be studied by the method of Poincaré. We consider the stance-compression to stance-decompression switching surface,  $S_{sc \rightarrow sd}$ , as a Poincaré section, and define the Poincaré map  $P : S_{sc \rightarrow sd} \rightarrow S_{sc \rightarrow sd}$ . Using this Poincaré map, we can numerically calculate the eigenvalues of its linearization about the fixed point. Numerical analysis shows that the obtained running gait has a dominant eigenvalue of 1.1928 and is unstable. Thus, an additional controller needs to be designed to stabilize the

running fixed point.

An outer-loop discrete event-based linear controller can be designed to stabilize the discrete linear system representing the linearized Poincaré map, as was done for Thumper in [17]. We identify certain parameters that can be varied step-to-step, and which could possibly affect stability of the fixed point. We choose the following parameters to be varied step-to-step: the stiffness and rest position parameters for the virtual compliance for the stance-compression and stance-decompression subphases, the touchdown angle, the torso offset and finally a parameter to change the flight duration. The linearized Poincaré map is obtained numerically and discrete LQR is used to find a feedback,  $\Gamma^\beta$ , that stabilizes the fixed point of the Poincaré map. On carrying out this procedure, we obtain a dominant eigenvalue of 0.8383, which shows that the fixed point is locally exponentially stabilized with this controller.

Next, prior to experimental validation, we study the robustness of the controller to perturbations. This controller can reject an error in torso of up to  $6^\circ$  in both directions, which is fairly good robustness to perturbations in torso angle. However, the controller is unable to reject an error in the form of the stance leg shape being bent by an additional  $5^\circ$ . Thus, there is a need for a controller that can improve the domain of attraction of the fixed point. This will be crucial for experimental validation.

The outer-loop  $\Gamma^\gamma$  controller is a heuristic nonlinear controller based on insight into simple models. For instance, on landing on a bent knee, the virtual compliance can be stiffened to prevent the stance leg from collapsing, thereby improving robustness to perturbations in the impact value of the stance leg shape. This outer-most controller is highly dependent on the morphology of the system and exists only to improve the robustness to perturbations in an experimental setting. The stability of the fixed point under the action of  $\Gamma^\gamma$  can once again be studied by the method of Poincaré by sampling the closed-loop hybrid system with the outer-loop  $\Gamma^\beta$  controller on a suitable Poincaré Section. Performing this numerically, a dominant eigenvalue of 0.6072 is obtained ensuring that the closed-loop system is still stable. Further details can be found in [21, Ch. VI, Ch. VIII], and [23].

## IV. EXPERIMENTAL VALIDATION OF THE RUNNING CONTROLLER

The running controller of Section III created stable running motions. This section documents experimental implementation of this controller on MABEL.

Before proceeding to experimental deployment, the proposed controller is tried on a detailed model developed in [15]. The detailed model introduces stretchy cables, compliant ground, and a more realistic model of the boom. This is a high DOF model and cannot be used for control design since an optimization process on this model is not computationally tractable. An important characteristic of the experimental system not captured by the model of Section II is highlighted. Specifically, MABEL has a cable-driven transmission, and these cables stretch. On running motions, there is severe





Fig. 8. A typical running step for MABEL. Snapshots are at an interval of 100 ms. A video of the running experiment is available on YouTube [22].

cable stretch along the leg shape direction, accounting for nearly 75% of motion in the stance knee at peak cable stretch on certain aggressive take-offs. The model of Section II-A assumed no cable stretch and the running controller needs to be modified to account for this discrepancy.

The cable stretch was identified in [15] and appears as an additional compliant element in series with the physical compliance. Since the running controller uses active force control in the stance phase for creating a virtual compliant element in series with the physical compliance, three sources of compliance (physical springs, cable stretch, virtual compliance) occur in series. Thus, the virtual compliance can be modified in such a way such that the effective compliance, after taking the cable stretch into account, has the stiffness that was initially designed for in the absence of cable stretch.

With this modification, the running controller induced stable running at an average speed of 1.95 m/s, and a peak speed of 3.06 m/s. Running speed is measured with respect to the center point of the hip between the two legs. A video of the experiment is available on YouTube [22]. 113 running steps were obtained and the experiment terminated when

the power to the robot was cut off. At 2 m/s, the average stance and flight times of 233 ms and 126 ms are obtained, respectively, corresponding to a flight phase that is 35% of the gait. At 3 m/s, the average stance and flight times of 195 ms and 123 ms are obtained respectively, corresponding to a flight phase that is 39% of the gait. An estimated ground clearance of 7.5–10 cm is obtained. Figure 8 depicts snapshots at 100 ms intervals of a typical running step.

## V. CONCLUSION

A control design based on virtual constraints and the framework of hybrid zero dynamics has been presented to create a compliant and actuated hybrid zero dynamics. An active force control strategy has been implemented within the compliant hybrid zero dynamics. Discrete-event-based control has been employed to create hybrid invariance, exponentially stabilize the periodic gait, and increase the domain of attraction of the periodic gait. The resulting controller has been successfully validated in experiments on MABEL achieving running at an average speed of 1.95 m/s, and a peak speed of 3.06 m/s.

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